

CIL
EMU CRITICAL ITEMS LIST

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12/24/93 SUPERSEDES 12/24/91

ANALYST:

NAME	P/N	FAILURE	CAUSE	FAILURE EFFECT	RATIONALE FOR ACCEPTANCE
CAUTION AND WARNING SYSTEM SWITCH, ITEM 364	2/2	368PM07: Switch jams in the proceed position or proceed contacts weld.		END ITEM: Unable to display CWS status information. GFE INTERFACE: Loss of capability to display IBM data, Loss of capability to access status.	A. Design - Switch mechanism and contacts enclosed in a hermetically sealed case backfilled with dry nitrogen. The switch is designed to withstand a toggle force of 25 lbs. without degradation in subsequent performance. Contact is accomplished through a roller type contact. This keeps switch forces to a minimum. The toggle/case interface is accomplished through a welded bellows which keeps switching forces to a minimum. B. Test - Component Acceptance Test - Vendor acceptance tests include 500 actuation cycles, contact resistance, and dielectric withstanding voltage tests. In-process Test - Switch operation and continuity are verified during in-process tests during DCM Item 350 assembly. PQA Test - Proper operation is verified during DCM PQA which includes continuity, functional tests, and operating torque. The switch is vibrated and exposed to thermal cycles during PQA as part of the DCM. Certification Test - The item completed the 15 year structural vibration and shock cert requirement during 10/83. The item was cycle certified by completed 127,800 cycles during 8/85. No Class I engineering changes have been issued since this configuration was certified. C. Inspection - To preclude failure due to internal contamination, the switches are assembled by the vendor in a Class 100,000 clean room. Flame switches are flushed internally using chlorofluorocarbon 11 and Benesolve D to remove contaminants prior to case welding. After welding, the switches are vacuum baked and backfilled with O2 to a pressure of 3-5 psig and sealed. Leak checks are performed during subsequent processing to verify seal integrity. X-ray inspections are performed, prior to run-in cycling and after vibration, to
SV767792-2 (1)			CAUSE: Contact weld caused by arcing or a failure of the hermetic seal and exposure to vacuum, jamming; shorting due to contamination.	MISSION: Terminate EVA.	

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ANALYST:

NAME	FAILURE MODE & CAUSES	FAILURE EFFECT	RATIONALE FOR ACCEPTANCE
2/2	360FM07:		verify absence of weld splatter and loose pieces, and to verify alignment.

D. Failure History -
None.

E. Ground Turnaround -
Tested per FEMU-R-001, Transducer and DGN Gage Calibration Check.

F. Operational Use -
Crew Response - PreEVA: If detected during EMU checkout or programmed leak check, discontinue use of EMU. Use third EMU if available.
EVA: When detected during periodic status check, troubleshoot using RIOB. Terminate EMU.
Training - Standard EMU training covers this failure mode.
Operational Considerations - EVA checklist procedures verify hardware integrity and systems operational status prior to EVA. Flight rules define go/no go criteria related to EMU CWS. Real Time Data System allows ground monitoring of EMU systems.