

CIL
EMU CRITICAL ITEMS LISTPage 1
Date: 11/10/94

12/24/94 SUPERSEDES 12/24/92

ANALYST:

NAME	FAILURE	FAILURE EFFECT	RATIONALE FOR ACCEPTANCE
P/N	MODE &		
QTY	CRIT	CAUSES	
POWER MODE SELECTOR SWITCH, ITEM 364	2/2	364FM07c Electrical open at power switch in battery position terminal (12). CAUSE: Cold solder joint, severed lead wire, contamination on contact.	<p>END ITEM: Connection between battery power and EVC lost. Loss of battery power connection to CLIV and Feedwater Valve.</p> <p>DTE INTERFACE: Cannot power EVC when on battery power. Cannot change positions of CLIV or Feedwater Valve when on battery power.</p> <p>MISSION: Terminate EVA, loss of EVC capabilities for EMU.</p> <p>CREW/VEHICLE: None.</p> <p>A. Design - Each of the three switches is sealed in a dry nitrogen filled hermetically sealed case. The switches are per MIL-S-8805/46 with the 10 amp contacts silver plated. Switch contacts rated for 10 ampere. Actual current flow is 3.8 ampere. The switch is designed to withstand a toggle force of 25 lbs. without degradation in subsequent performance. The ball socket of the toggle pivot is greased (Graycote 601) prior to assembly.</p> <p>Hieroswitch actuator overtravel is adjusted to .007 inch minimum to ensure the common contact arm rotates completely over to the normally open contact.</p> <p>B. Test - Component Acceptance Test - Switch operation and continuity are verified during vendor acceptance tests. The switch is also subjected to 500 run-in cycles and an axial pull test on the handle to verify that it will not come loose during normal use.</p> <p>In-Process Test - Operation and integrity of the switch are verified during four separate In-process tests during initial Item 350 assembly. These tests include continuity and output voltage. The switch is cycled during these tests.</p> <p>PQA Test - The switch is subjected to Acceptance/PQA testing as part of Item 350. Tests include continuity, operating torque, vibration, thermal cycling, and thermal vacuum. The switch is also cycled during Item 350 Acceptance/PQA electrical functional tests.</p> <p>Certification Test - The item completed 5,464 inductive and 8,536 resistive cycles during 1/81 which fulfilled the cycle certification requirement of 5,464 and 8,536 respectively. Class I Engineering Change 42806-366 (Toggle Handle Pull Test) has been incorporated since this configuration was certified on 1/84 per SEMU-560.</p>
SU770596-4	(1)		

CIL
EMU CRITICAL ITEMS LIST

Page: 2
Date: 11/10/94

12/24/94 SUPERSEDES 12/26/92

ANALYST:

NAME	P/N	QTY	CRIT	FAILURE MODE & CAUSES	FAILURE EFFECT	RATIONALE FOR ACCEPTANCE
			2/2	364FM07:		<p>E. Inspection -</p> <p>To preclude failure due to internal contamination, the switches are assembled by the vendor in an environmentally controlled room. Assembly and processing is per MIL-S-880566. The switches receive in-process cycling and leak checks. The entire Item 364 is x-ray inspected for acceptability of brazing.</p> <p>The solder terminals on the switch are visually checked as part of source inspection for the part. The terminals are also inspected after lead wires are soldered on during DCM assembly. Solder joints are inspected per NM8 5300.4 (5A-1).</p> <p>D. Failure History -</p> <p>J-EMU-360-006 (10-18-83)</p> <p>The BITE light failed to turn on upon power switchover during PIA tests. The outage was found to be caused by a mechanical failure of the Power Mode Switch (364) which prevented proper power switchover. EC 42806-386 added a pull test to the 364 vendor tests to insure the switch toggle arm would not come loose during normal use. This EC created the -2 switch configuration. Certified on 1/84 per SEMI-S60.</p> <p>E. Ground Turnaround -</p> <p>Switches are tested during FEMU-A-001, EMU Vacuum Chamber Run, EMU Checkout in Orbiter, Orbiter Power Interface and SEMU Comm & Biomed check.</p> <p>F. Operational Use -</p> <p>Crew Response - PreEVA: Troubleshoot problem. If no success, consider third EMU if available. Otherwise, EMU go for SDU standby.</p> <p>EVA: Terminate EVA.</p> <p>Training - Standard training covers this failure mode.</p> <p>Operational Considerations - EVA checklist procedures verify hardware integrity and systems operational status prior to EVA. Flight rules require that EVA be terminated if two-way communication between each EV crewmember and orbiter, either direct or through relay, is unavailable.</p> <p>Real Time Data System allows ground monitoring of EMU</p>

EMU - 1548

CIL
EMU CRITICAL ITEMS LIST

Page: 3
Date: 11/10/94

12/24/94 SUPERSEDES 12/24/92

ANALYST:

NAME	FAILURE	MODE &	FAILURE EFFECT	RATIONALE FOR ACCEPTANCE
P/N		CAUSES		
QTY	CRIT			
2/2	364PN07:			systems.