

FAILURE MODES EFFECTS ANALYSIS (FMEA) -- NON-CIL HARDWARE
NUMBER:05-3A-B26-1 -X

SUBSYSTEM NAME: MULTIFUNCTION ELECTRONIC DISPLAY SUBSYSTEM
REVISION: 0 01/19/95

PART DATA

| | PART NAME | PART NUMBER |
|-----|--------------------|----------------------|
| | VENDOR NAME | VENDOR NUMBER |
| LRU | : PANEL C2A2 | VO70-73D280 |
| LRU | : PANEL R12A2 | VO70-730335 |
| SRU | : SWITCH, TOGGLE | ME452-0102-7106 |

EXTENDED DESCRIPTION OF PART UNDER ANALYSIS:
 SWITCH, "CRT" MAJOR, FUNCTION SELECT 1P3P "GNC-SM-PL" (GUIDANCE NAVIGATION AND CONTROL-SYSTEM MANAGEMENT-PAYLOAD)

REFERENCE DESIGNATORS: 35V73A2A2S2
 35V73A2A2S3
 35V73A2A2S4
 32V73A12A2S2

QUANTITY OF LIKE ITEMS: 4
 THREE FRONT
 ONE AFT

FUNCTION:
 PROVIDES MEANS FOR CONTROLLING AND SELECTING THE IDP, OR "CRT" MDU MAJOR FUNCTION. THE SWITCH INTERFACES WITH THE GENERAL PURPOSE COMPUTER (GPC) THROUGH THE INTEGRATED DISPLAY PROCESSOR (IDP).

REFERENCE DOCUMENTS: VS70-730182D
 SSD90D0009B, CP#1
 MC409-0185D, AMENDMENT E01
 SSD92D0843D, CP#2

FAILURE MODES EFFECTS ANALYSIS FMEA -- NON-CIL FAILURE MODE

NUMBER: 05-3A-B26-1-02

REVISION#: 1 04/26/98

SUBSYSTEM NAME: MULTIFUNCTION ELECTRONIC DISPLAY SUBSYSTEM

LRU: PANEL C2A2, R12A2

CRITICALITY OF THIS

ITEM NAME: SWITCH, TOGGLE

FAILURE MODE: 1R3

FUNCTIONAL CRITICALITY/**REQUIRED FAULT TOLERANCE/ACHIEVED FAULT TOLERANCE: 1R/2/3****FAILURE MODE:**

FAILS CLOSED, PREMATURE CLOSURE, CONTACT-TO-CONTACT SHORT

MISSION PHASE:

| | |
|----|----------------|
| PL | PRE-LAUNCH |
| LO | LIFT-OFF |
| OO | ON-ORBIT |
| DO | DE-ORBIT |
| LS | LANDING/SAFING |

VEHICLE/PAYLOAD/KIT EFFECTIVITY:

| | |
|-----|-----------|
| 102 | COLUMBIA |
| 103 | DISCOVERY |
| 104 | ATLANTIS |
| 105 | ENDEAVOUR |

CAUSE:

CONTAMINATION, MECHANICAL SHOCK, VIBRATION, PROCESSING ANOMALY, PIECE PART STRUCTURAL FAILURE

CRITICALITY 1/1 DURING INTACT ABORT ONLY? NO**CRITICALITY 1R2 DURING INTACT ABORT ONLY (AVIONICS ONLY)? NO**

REDUNDANCY SCREEN

| |
|---------|
| A) PASS |
| B) PASS |
| C) PASS |

PASS/FAIL RATIONALE:

A)

B)

C)

METHOD OF FAULT DETECTION:

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VISUAL

MASTER MEAS. LIST NUMBERS: V72X5643B
V98J4349C
V73K2003E
V73K2004E
V73K2005E
V72X5713B
V98J4359C
V73K2013E
V73K2014E
V73K2015E
V72X5803B
V98J4369C
V73K2023E
V73K2024E
V73K2025E
V72X5903B
V98J4379C
V73K2053E
V73K2054E
V73K2055E

CORRECTING ACTION: MANUAL

CORRECTING ACTION DESCRIPTION:
MDU'S COMMANDED BY OTHER IDP'S ARE AVAILABLE FOR DISPLAY.

REMARKS/RECOMMENDATIONS:
MDU'S COMMANDED BY OTHER IDP'S PROVIDE ADEQUATE BACKUP. IDP WILL DEFAULT TO GNC FUNCTION IF SWITCH IS IN INDETERMINANT POSITION.

- FAILURE EFFECTS -

(A) SUBSYSTEM:
SWITCH REMAINS IN THE PREVIOUSLY COMMANDED MAJOR FUNCTION; INABILITY TO TRANSFER TO GN&C MODE OR OTHER MODES. AN ALTERNATE IDP WILL HAVE TO BE SELECTED FOR MAJOR FUNCTION RELATED KEYBOARD INPUTS TO THE GPC. THE MULTIFUNCTION DISPLAY UNIT (MDU) CONTINUES TO UPDATE DATA OF THE PREVIOUS COMMANDED MAJOR FUNCTION.

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(B) INTERFACING SUBSYSTEM(S):

NO EFFECT FIRST FAILURE

(C) MISSION:

NO EFFECT FIRST FAILURE

(D) CREW, VEHICLE, AND ELEMENT(S):

NO EFFECT FIRST FAILURE

(E) FUNCTIONAL CRITICALITY EFFECTS:

POSSIBLE LOSS OF CREW/VEHICLE AFTER THREE FAILURES (SWITCH FAILS CLOSED WILL RESULT IN IDP STUCK IN PREVIOUS COMMANDED MAJOR FUNCTION, LOSS OF THE OTHER TWO IDP'S) DUE TO LOSS OF ABILITY TO ADVISE ORBITER CONTROL SYSTEMS TO ACCEPT UPDATED STATE VECTOR DATA, NAVIGATIONAL AND AIR DATA INPUTS REQUIRED TO ASSURE SAFE VEHICLE CONTROL, AND MANUAL OPS MODE TRANSITIONS.

- TIME FRAME -

TIME FROM FAILURE TO CRITICAL EFFECT: MINUTES

TIME FROM FAILURE OCCURRENCE TO DETECTION: IMMEDIATE

TIME FROM DETECTION TO COMPLETED CORRECTING ACTION: SECONDS

**IS TIME REQUIRED TO IMPLEMENT CORRECTING ACTION LESS THAN TIME TO EFFECT?
YES**

**RATIONALE FOR TIME TO CORRECTING ACTION VS TIME TO EFFECT:
N/A (CORRECTIVE ACTION CAN BE COMPLETED BEFORE CRITICAL EFFECT)**

HAZARD REPORT NUMBER(S):

HAZARD(S) DESCRIPTION:

- APPROVALS -

SS&PAE ENGR

: N. D. NGUYEN

[Handwritten Signature]

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MEDS SYSTEM
MEDS HARDWARE

: M. B. WARNER
: R. M. SITAPARA

: *M. B. Warner*
: *Ramon M. Sitapara 4/28/98*