

SHUTTLE CRITICAL ITEMS LIST - ORBITER

SUBSYSTEM : ACTIVE THERMAL CONTROL FMEA NO 06-3C -0220 -1 REV:08/26/
 ASSEMBLY : FREON THERMAL LOOP CRIT. FUNC: :
 P/N RI : MC250-0001-0270 CRIT. HDW:
 P/N VENDOR: SV729792-2 VEHICLE 102 103 104
 QUANTITY : 2 EFFECTIVITY: X X X
 : TWO, ONE PER LOOP PHASE(S): PL LO X OO X DO X LS

PREPARED BY: DES O. TRAN *O. Tran* APPROVED BY: *[Signature]* REDUNDANCY SCREEN: A-PASS B-PASS C-PAS
 REL D. RISING *D. Rising* DES *[Signature]* APPROVED BY (NASA) *[Signature]*
 QE W. SMITH *W. Smith* SSM *[Signature]* REL *[Signature]*
 QE *[Signature]* QE *[Signature]*

ITEM:
 VALVE MODULE, FLOW PROPORTIONAL.

FUNCTION:
 THE VALVE MODULE PROPORTIONS THE FLOW OF FREON TO THE PAYLOAD HEAT EXCHANGER AND THE WATER/FREON INTERCHANGER.

FAILURE MODE:
 EXTERNAL LEAKAGE.

CAUSE(S):
 CORROSION, VIBRATION, MECHANICAL SHOCK.

EFFECT(S) ON:
 (A) SUBSYSTEM (B) INTERFACES (C) MISSION (D) CREW/VEHICLE
 (A,B) POSSIBLE LOSS OF ONE FREON COOLANT LOOP FOR VEHICLE COOLING.
 (C) POSSIBLE LOSS OF MISSION. EARLY MISSION TERMINATION FOR LOSS OF FREON LOOP.
 (D) SECOND ASSOCIATED FAILURE (LOSS OF REDUNDANT FREON COOLANT LOOP) WILL CAUSE LOSS OF ALL VEHICLE COOLING AND MAY RESULT IN LOSS OF CREW/VEHICLE.

DISPOSITION & RATIONALE:
 (A) DESIGN (B) TEST (C) INSPECTION (D) FAILURE HISTORY (E) OPERATIONAL USE

(A) DESIGN
 THE VALVE CONSISTS OF A STAINLESS STEEL HOUSING AND SPOOL. THE SPOOL HAS 8 DIFFERENT SIZED ORIFICES - THE SMALLEST IS 0.062 INCHES. 25 MICRON (ABS) FILTERS AT THE INLET AND OUTLET OF THE VALVE PROTECT AGAINST CONTAMINATION. MATERIALS USED ARE CORROSION RESISTANT AND COMPATIBLE WITH FREON 21. DESIGN PROOF PRESSURE IS 1.5 AND BURST PRESSURE IS 2.0 TIMES THE MAXIMUM OPERATING PRESSURE. REDUNDANT DYNAMIC KALREZ SEALS ARE USED TO SEAL BETWEEN THE FREON AND AMBIENT. THE MODULE IS BRAZED INTO THE PCL SYSTEM.

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(B) TEST

QUALIFICATION TEST - QUALIFICATION TESTED FOR 100 MISSION LIFE. VIBRATION TESTED AT 2.0 G²/HZ FOR 84 MIN/AXIS, SHOCK TESTED AT +/- 20 G EACH AXIS. THE VALVE WAS CYCLED 1000 TIMES WITH NO FAILURES OF THIS TYPE.

ACCEPTANCE TEST - COMPONENT ATP INCLUDES LEAK CHECK.

OMRSD - PCL ARE LEAK CHECKED PRIOR TO EACH FLIGHT. FREON CHEMICAL ANALYSIS PER SE-S-0073 DURING SERVICING. VEHICLE FREON IS SERVICED THROUGH A 10 MICRON (ABS) GSE FILTER.

(C) INSPECTION

RECEIVING INSPECTION

RAW MATERIAL AND PURCHASED COMPONENTS REQUIREMENTS ARE VERIFIED BY RECEIVING INSPECTION. COATING AND PLATING MATERIALS AND PROCESSES ARE VERIFIED BY INSPECTION.

CONTAMINATION CONTROL

FORMAL CONTAMINATION CONTROL PLAN IS VERIFIED BY INSPECTION. CONTAMINATION CONTROL PROCESSES AND CLEAN AREAS ARE VERIFIED BY INSPECTION. CORROSION PROTECTION PROVISIONS ARE VERIFIED BY INSPECTION. SYSTEM FLUID SAMPLES ARE PERIODICALLY ANALYZED FOR CONTAMINATION AND VERIFIED BY INSPECTION.

ASSEMBLY/INSTALLATION

PARTS PROTECTION, MANUFACTURING PROCESSES, INSTALLATION AND ASSEMBLY OPERATIONS ARE VERIFIED BY INSPECTION ON SHOP TRAVELERS. MEASUREMENT STANDARDS AND TEST EQUIPMENT IMPLEMENTATION PER REQUIREMENTS OF MIL SPECIFICATIONS ARE VERIFIED BY INSPECTION. TORQUE CERTIFICATION IS VERIFIED BY INSPECTION.

NONDESTRUCTIVE EVALUATION

LEAK TEST IS VERIFIED BY INSPECTION.

TESTING

FUNCTIONAL TEST IS MONITORED BY INSPECTION TO VERIFY FLOWRATE IS WITHIN SPECIFIED LIMITS.

HANDLING/PACKAGING

HANDLING, PACKAGING, AND STORAGE REQUIREMENTS ARE VERIFIED BY INSPECTION.

(D) FAILURE HISTORY

NO FAILURE HISTORY.

(E) OPERATIONAL USE

ON-BOARD ALARMS, FREON FLOW, WILL PROVIDE INDICATION OF HARDWARE FAILURE. FREON PUMP WILL BE TURNED OFF AND LOSS OF ONE FREON LOOP POWERDOWN WILL BE PERFORMED. ENTRY AT NEXT PRIMARY LANDING SITE.

06-3C-72