

SHUTTLE CRITICAL ITEMS LIST - ORBITER

SUBSYSTEM : LIFE SUPPORT FMEA NO 06-2E -0418 -1 REV:10/29/
 ASSEMBLY : VENT AND DUMP EQUIPMENT CRIT. FUNC: 1
 P/N RI : ME363-0043-0003/-0005 CRIT. HDW:
 P/N VENDOR: 66-2870-3 VEHICLE 102 103 104
 QUANTITY : 1 EFFECTIVITY: X X X
 : PHASE(S): PL LO OO X DO X LS
 : ONE PER VEHICLE

REDUNDANCY SCREEN: A-PASS B-PASS C-PA
 PREPARED BY: APPROVED BY: APPROVED BY (NASA):
 DES S. CASTILLO DES *[Signature]* SSM *[Signature]*
 REL L. SCHASCHL REL *[Signature]* REL *[Signature]*
 QE M. SAVALA QE *[Signature]* QE *[Signature]*

ITEM:
 NOZZLE, VACUUM VENT

FUNCTION:
 VENTS WASTE GASES FROM THE WASTE COLLECTOR, HYDROGEN FROM THE HYDROGEN SEPARATOR, AIR FROM THE AIRLOCK AND N2 FROM THE NITROGEN RELIEF VALVES

FAILURE MODE:
 RESTRICTED FLOW

CAUSE(S):
 CONTAMINATION

EFFECT(S) ON:
 (A)SUBSYSTEM (B)INTERFACES (C)MISSION (D)CREW/VEHICLE

(A) LOSS OF ONE AIRLOCK DEPRESSURIZATION PATH. POSSIBLE BUILDUP OF HYDROGEN.

(B, C) NO EFFECT.

(D) SECOND ASSOCIATED FAILURE (CABIN LEAK INTO DUCT), CAN PRESSURIZE DUCT WITH CABIN AIR (O2) AND CREATE AN EXPLOSIVE MIXTURE. AN EXPLOSION CAN RUPTURE VACUUM DUCT AND CAUSE LOSS OF CABIN AIR AND CREW/VEHICLE.

DISPOSITION & RATIONALE:
 (A)DESIGN (B)TEST (C)INSPECTION (D)FAILURE HISTORY (E)OPERATIONAL USE

(A) DESIGN
 THE 2 INCH DIAMETER FLANGED OUTLET IS CONSTRUCTED OF 21-6-9 CRES. THE 2 INCH OUTSIDE DIAMETER TUBING (0.035 INCH WALL THICKNESS) IS CONSTRUCTED OF 321 CRES. THE NOZZLE ASSEMBLY IS CLEANED AND PASSIVATED PER QQ-P-35. A PLUG PROTECTS THE DUCT AND NOZZLE AGAINST CONTAMINATION DURING GROUND OPERATIONS. TWO HEATER ELEMENTS. ALL OF THE MATERIALS USED ARE COMPATIBLE WITH THE WORKING FLUIDS.

(B) TEST
 QUALIFICATION TESTS FOR 100 MISSION LIFE INCLUDE: RANDOM VIBRATION INCREASING AT THE RATE OF 6 db/OCTAVE FROM 20 TO 90 HZ; CONSTANT AT 1 G SQ/HZ TO 300 HZ; DECREASING AT 6 DB/OCTAVE TO 2000 HZ, 34 MINUTES/AXIS,

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AND SHOCK TESTED AT 20 G PER AXIS.

ACCEPTANCE TESTS - PRESSURE TEST OF 60 PSID INTERNAL PRESSURE FOR 5 MINUTES.

OMRSD: FLOW THROUGH NOZZLE IS VERIFIED BEFORE EACH FLIGHT.

(C) INSPECTION

RECEIVING INSPECTION

MATERIAL IS VERIFIED BY PHYSICAL-CHEMICAL REPORTS AT RECEIVING INSPECTION.

CONTAMINATION CONTROL

CLEANLINESS LEVELS AND CONTAMINATION CONTROL PLAN ARE VERIFIED BY INSPECTION.

CRITICAL PROCESSES

BRAZING AND PASSIVATION PROCESSES ARE VERIFIED BY INSPECTION. TIG WELD SCHEDULES ARE VERIFIED BY INSPECTION.

ASSEMBLY/INSTALLATION

CRITICAL DIMENSIONS ARE VERIFIED BY INSPECTION. MANUFACTURING PROCESSES, INSTALLATION, AND ASSEMBLY ARE VERIFIED BY INSPECTION.

NONDESTRUCTIVE EVALUATION

MATERIAL DEFECTS ARE VERIFIED BY X-RAY INSPECTION.

TESTING

BRAZED JOINTS ARE SUBJECTED TO VISUAL INSPECTION AND LEAK TEST. WATER NOZZLE IS SUBJECTED TO STATIC PULL TEST FOR WIRE SEPARATION DURING ACCEPTANCE TEST. ACCEPTANCE TEST IS VERIFIED BY INSPECTION.

HANDLING/PACKAGING

PARTS PROTECTION IS VERIFIED BY INSPECTION.

(D) FAILURE HISTORY

NO FAILURES.

(E) OPERATIONAL USE

TO ISOLATE THE O2 FROM THE H2, CLOSE THE VACUUM VENT VALVE ON THE WASTE COLLECTOR SYSTEM (WCS) PANEL. FOR AIRLOCK DEPRESSURIZATION, USE OUTER HATCH EQUALIZATION VALVES.